

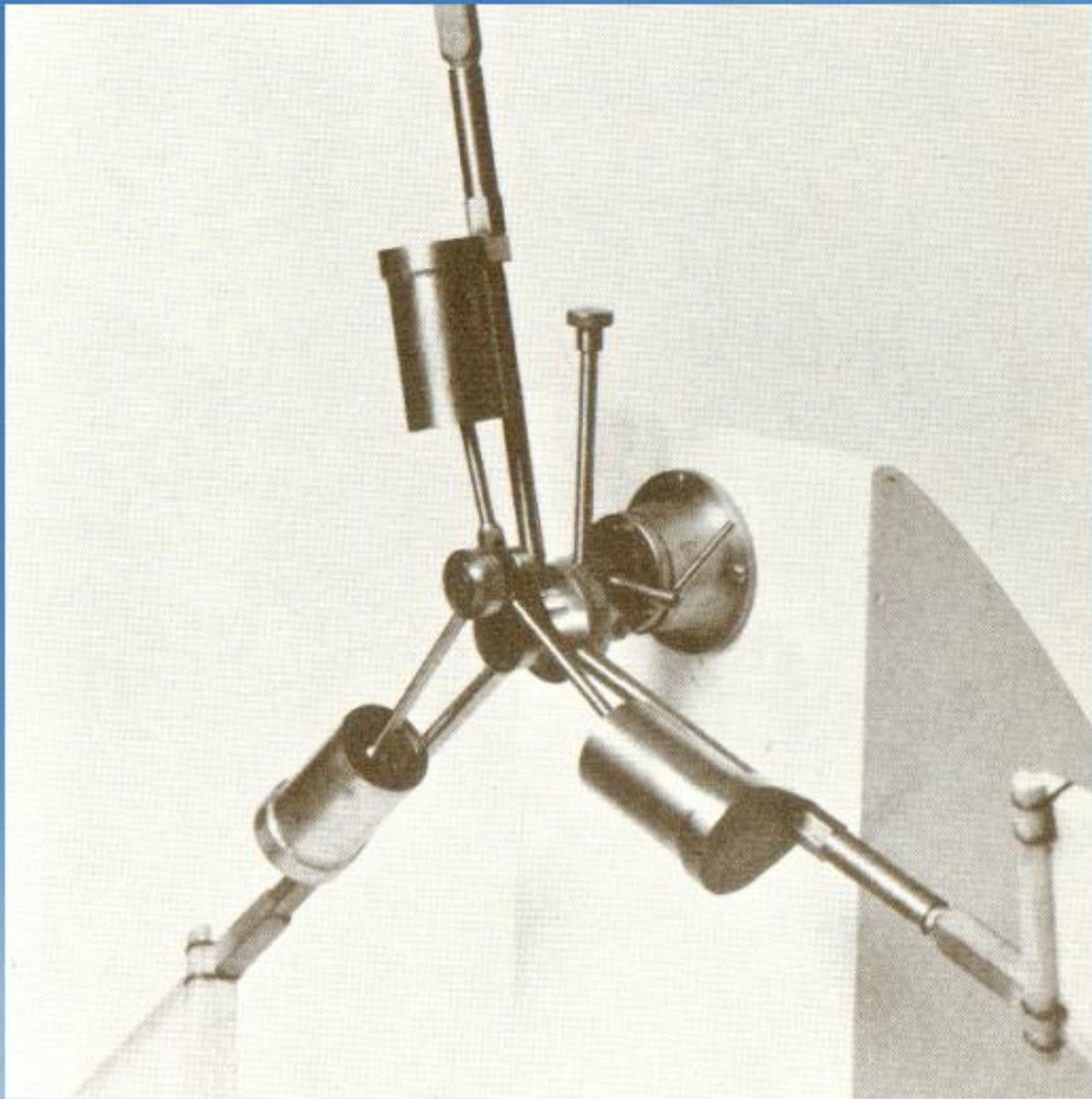
ICAS Daniel & Florence Guggenheim Memorial Lecture

**COLLABORATION CHALLENGES IN THE
GLOBAL AEROSPACE MARKET FOR SMALLER
COUNTRIES
– AN AUSTRALIAN PERSPECTIVE**

W.H. Schofield

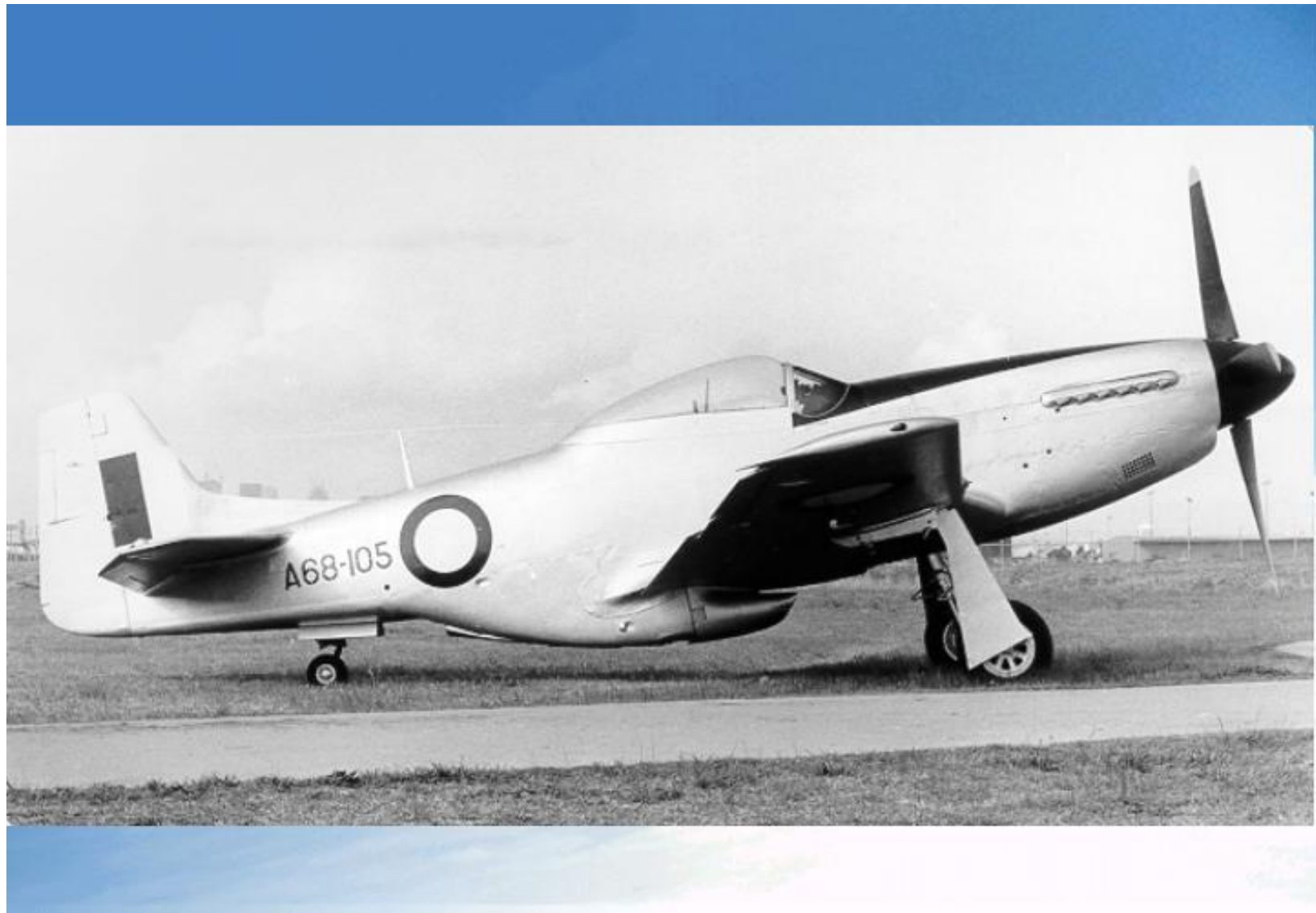
Cooperative Research Centre for Advanced Composite Structures
506 Lorimer Street, Fishermans Bend, Victoria, 3207, Australia









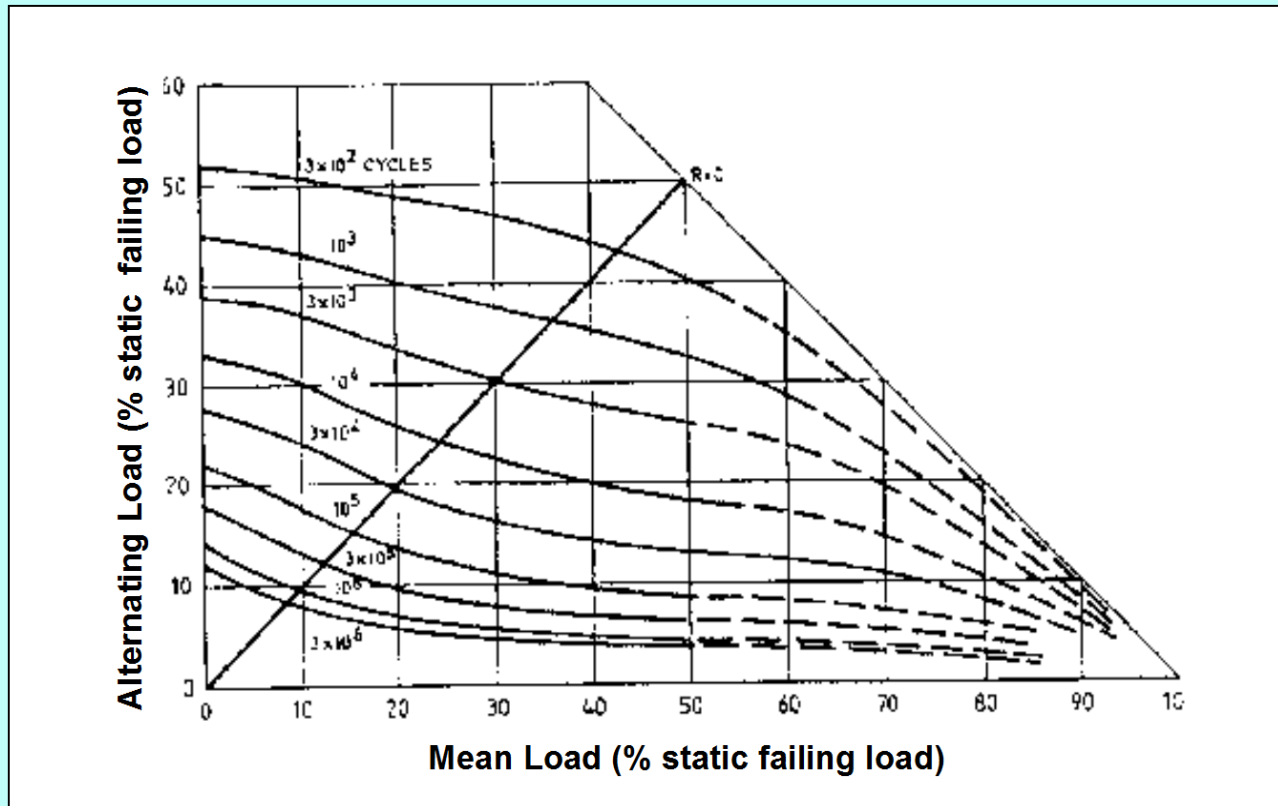




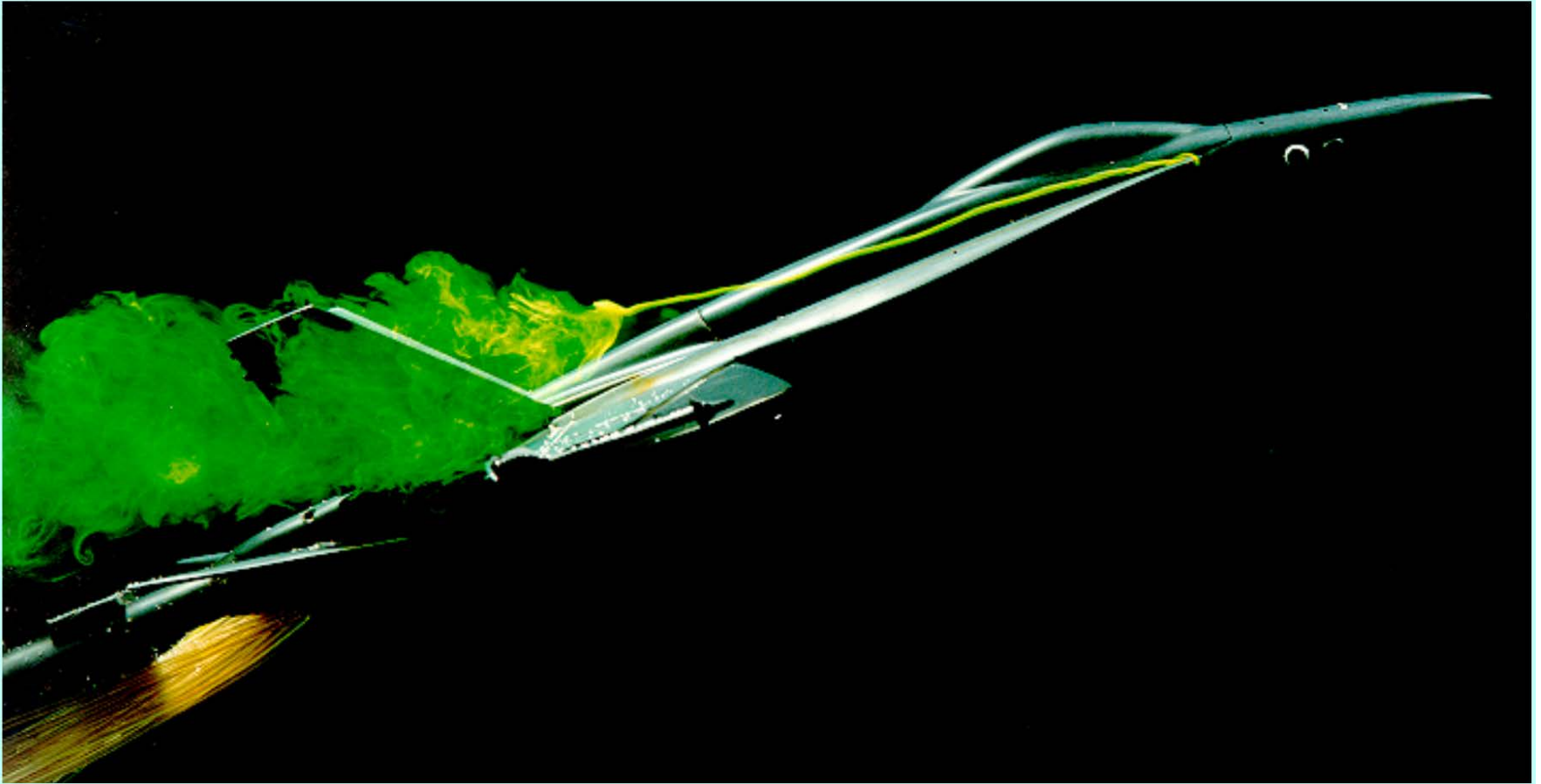
STINSON AIRCRAFT



Alternating Load - Mean Load Diagram for Mustang Wing

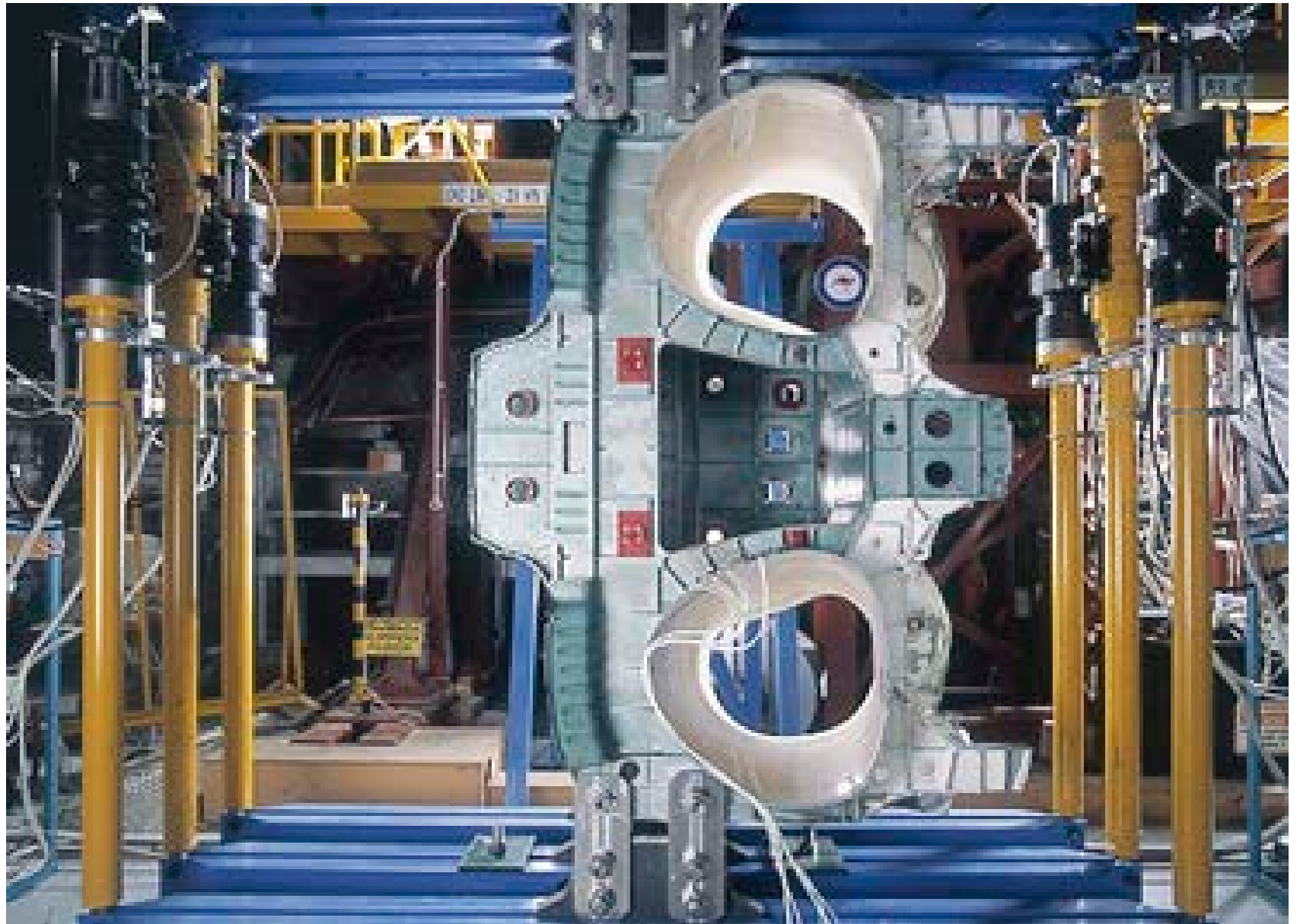


VORTEX BREAKDOWN ON AN F18 MODEL IN AMRL WATER TUNNEL





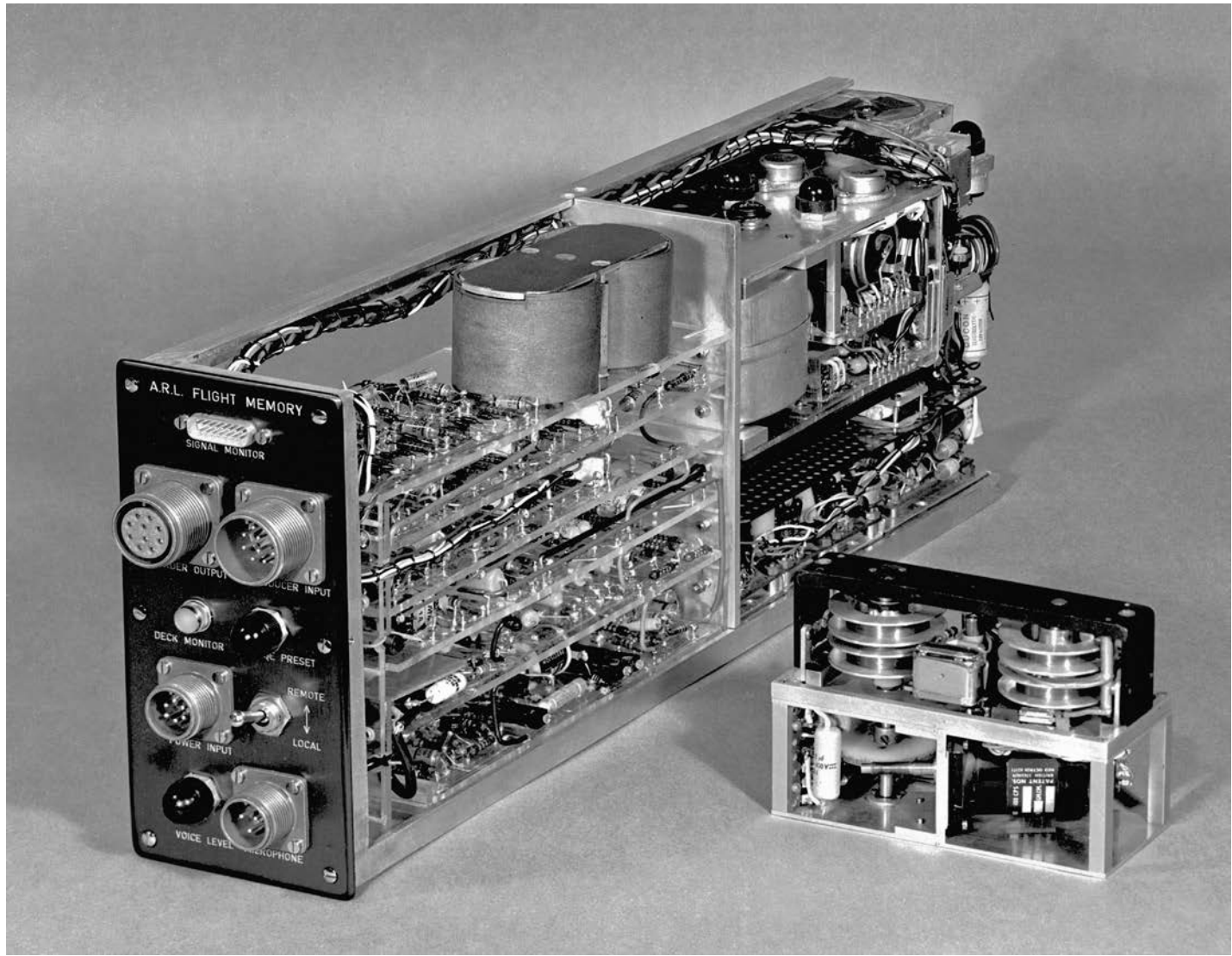




CASE STUDY: PC-9 FATIGUE TEST

- Aircraft interim fatigue life 6,000 hours
- Australian target fatigue life 12,000 hours
- Increase in fatigue life 100%
- Capital cost of PC9 fleet including ancillary costs (in 1996 dollars) \$438 million
- Value of additional life arising out of the test (100% of \$438m) \$438 million
- Estimated cost of DSTO/RAAF Program \$6.0 million
- Estimated cost of RAAF flight program to develop flight load data \$1.1 million

- **RETURN ON INVESTMENT 62:1 (or 6,200%)**

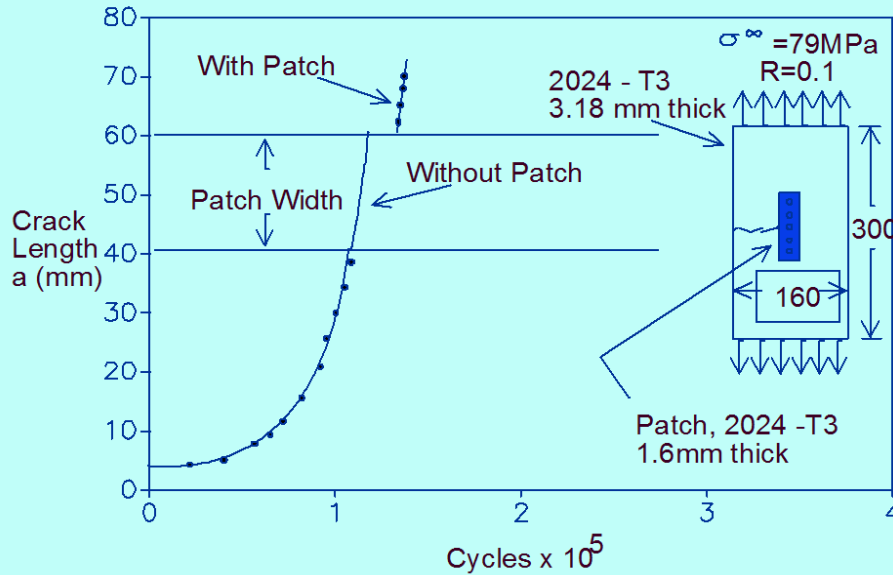




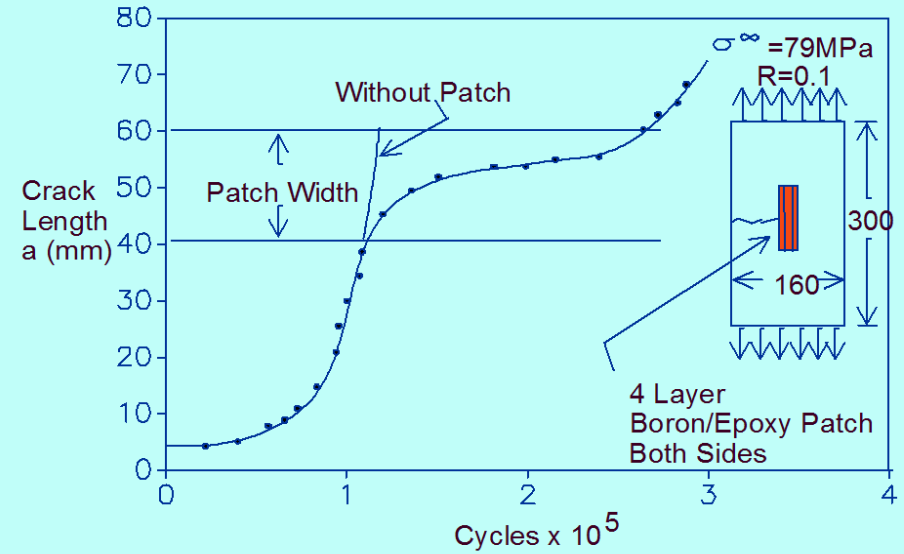








MECHANICAL REPAIR



COMPOSITE REPAIR

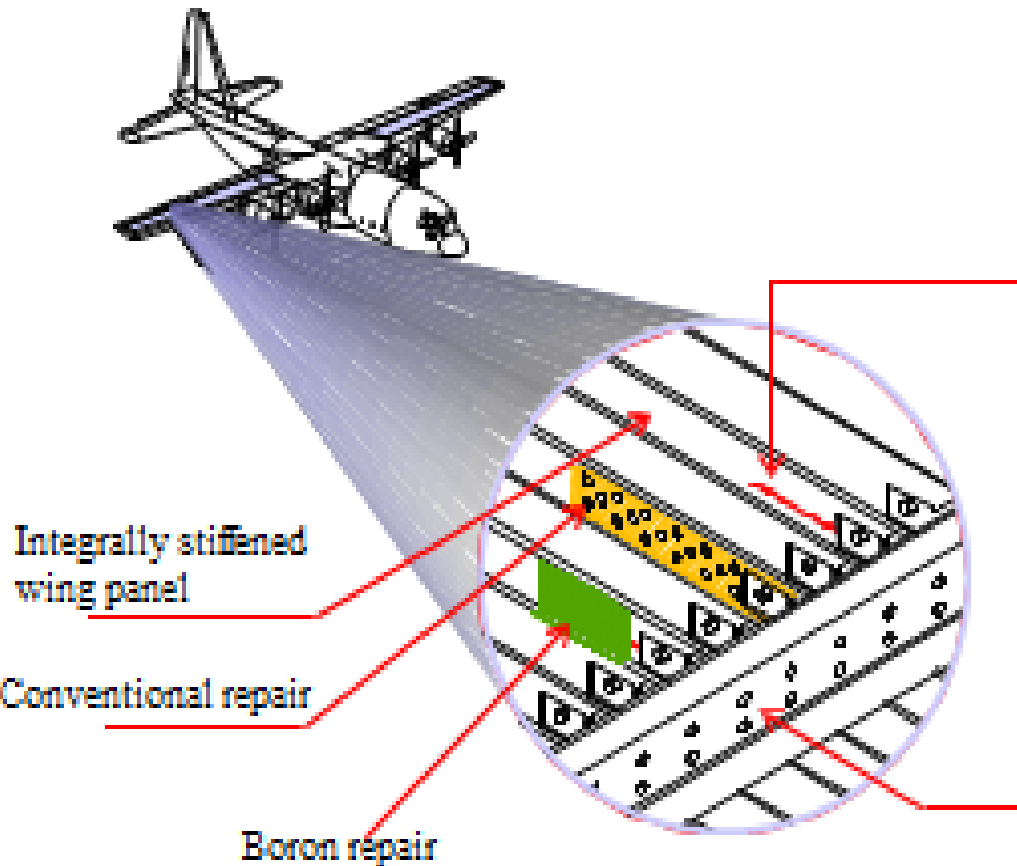


Stress Corrosion Crack Repair

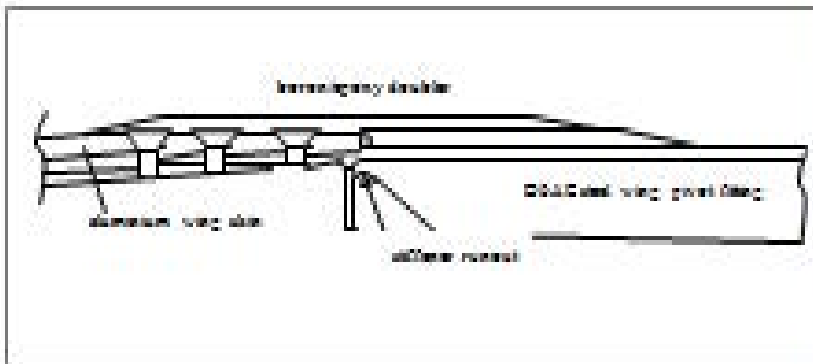
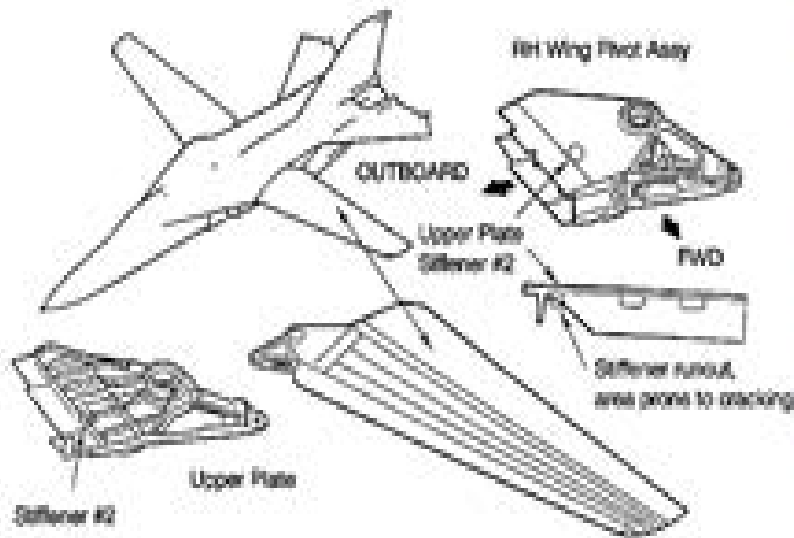
Solution

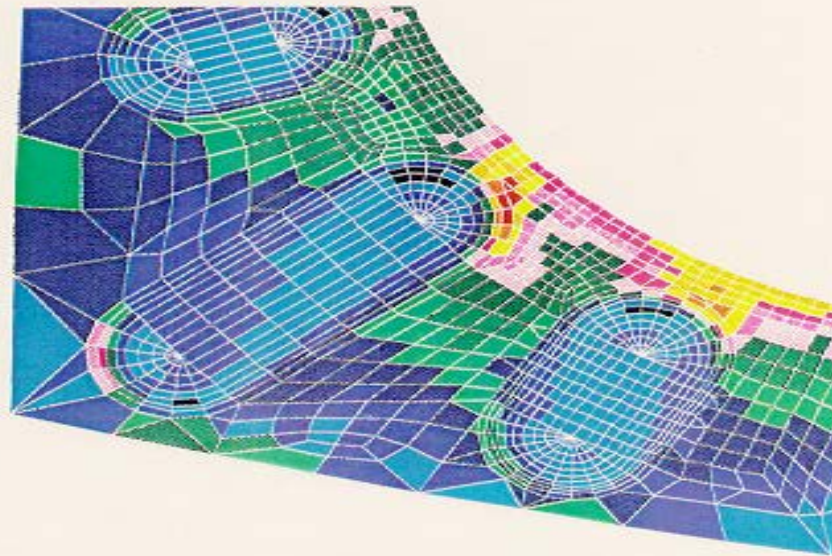
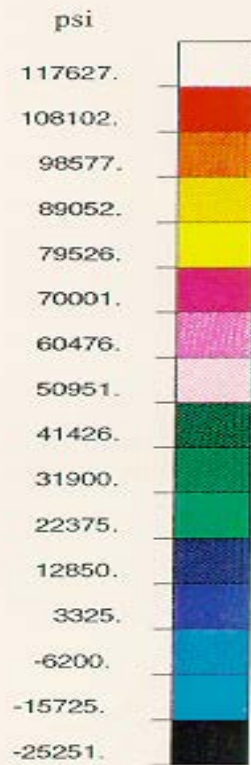
- boron/epoxy bonded repair applied inside the wet wing
- over 1000 repairs applied with no crack growth in 20 years
- estimated savings over \$100 million by 1985

Completed Repair

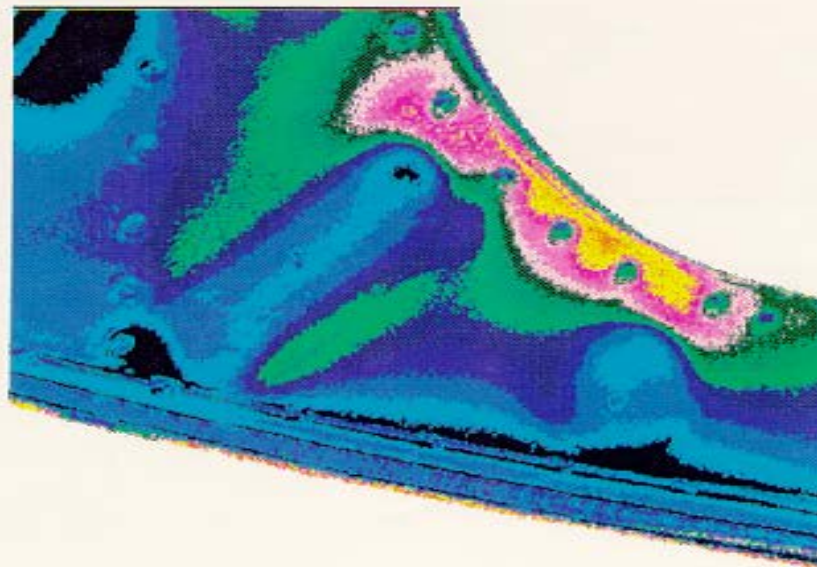


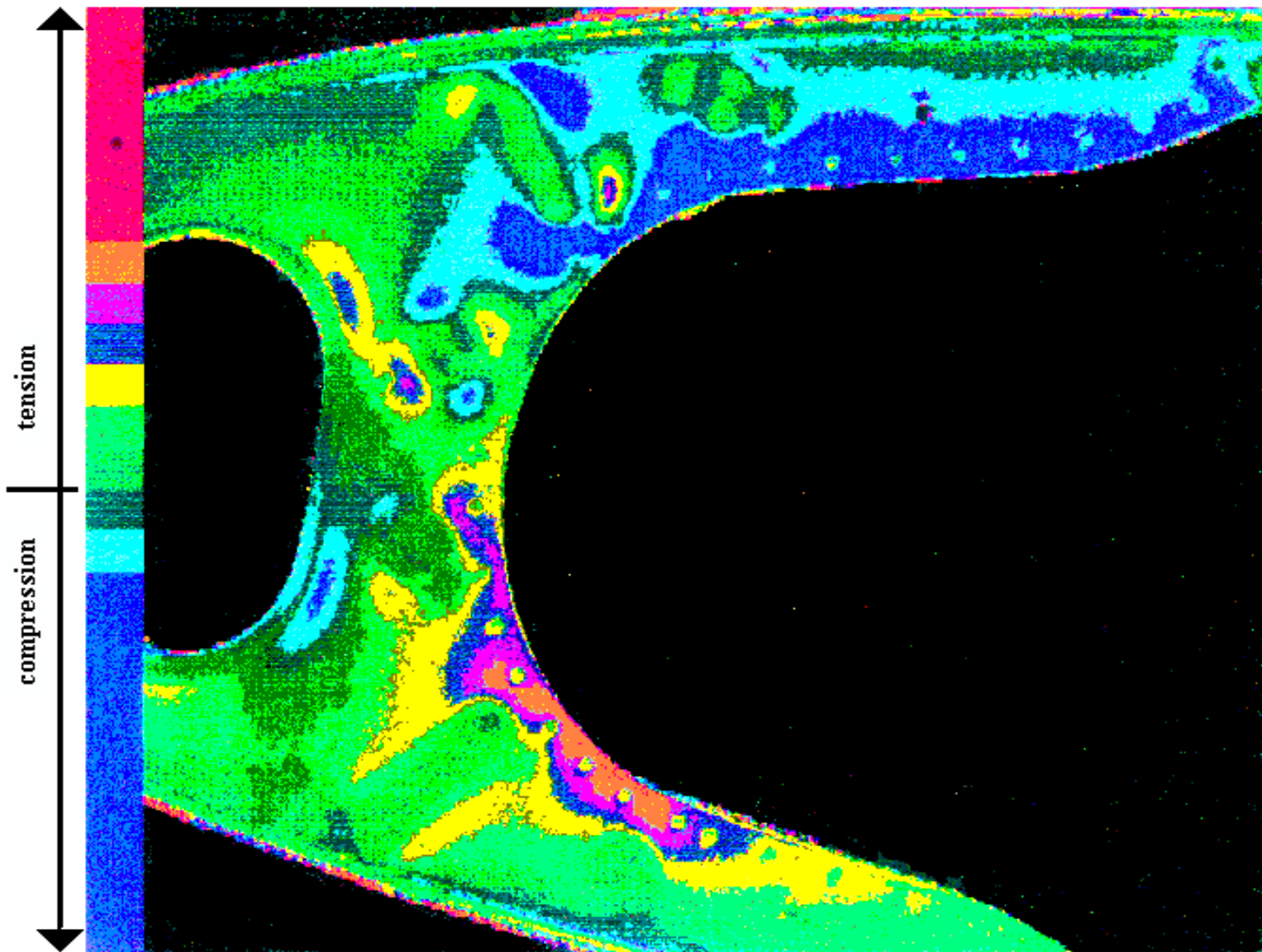
Strain Reduction: Wing Pivot Fitting F111



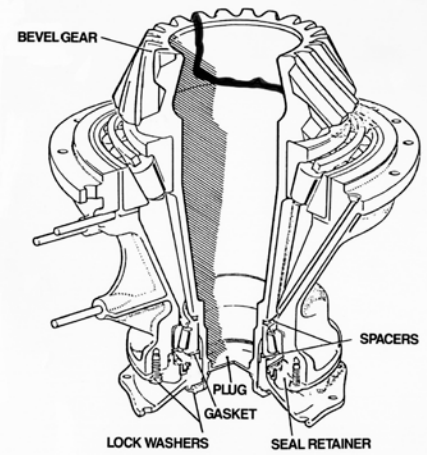
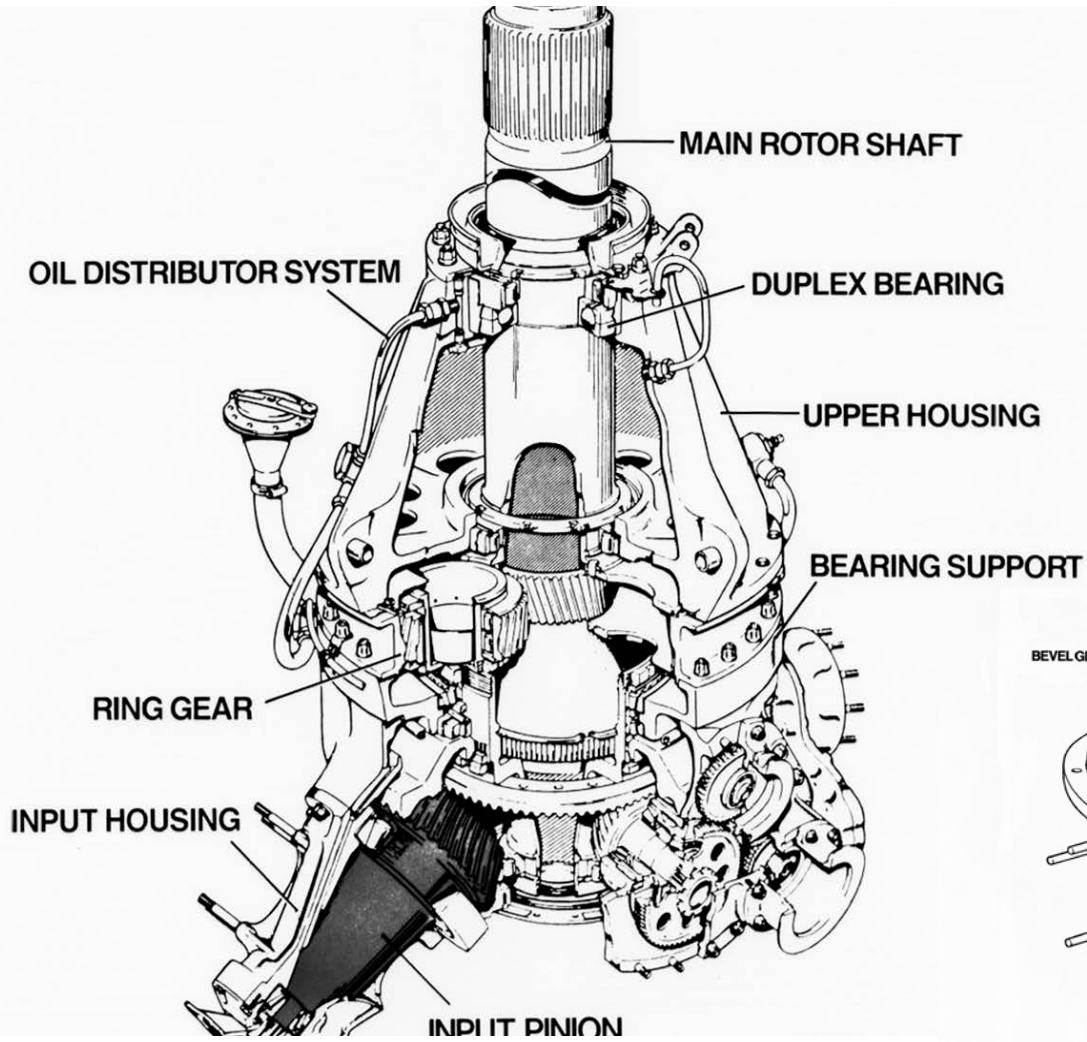


NAVAIR 3-D FE ANALYSIS OF CRITICAL REGION.





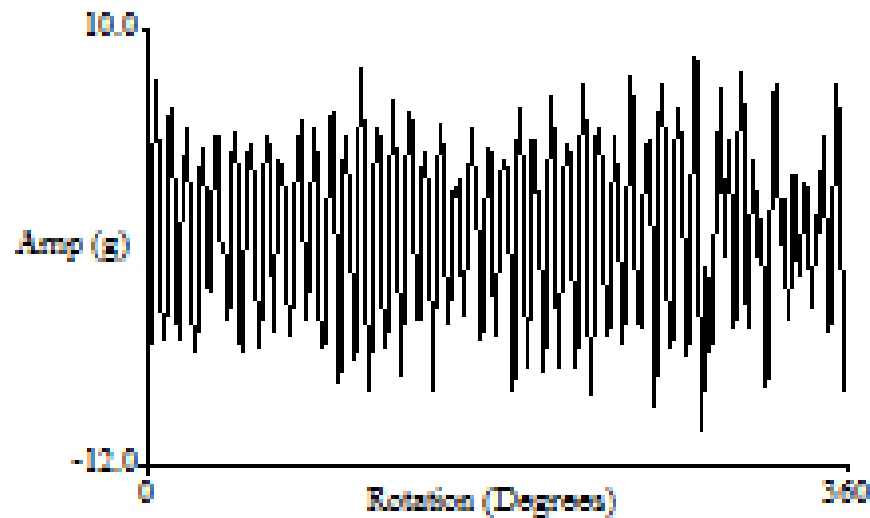
FAST STRESS SCAN OF LEADING EDGE
(NOTE STRESS CONCENTRATION EFFECTS OF STIFFENING DIMPLES).



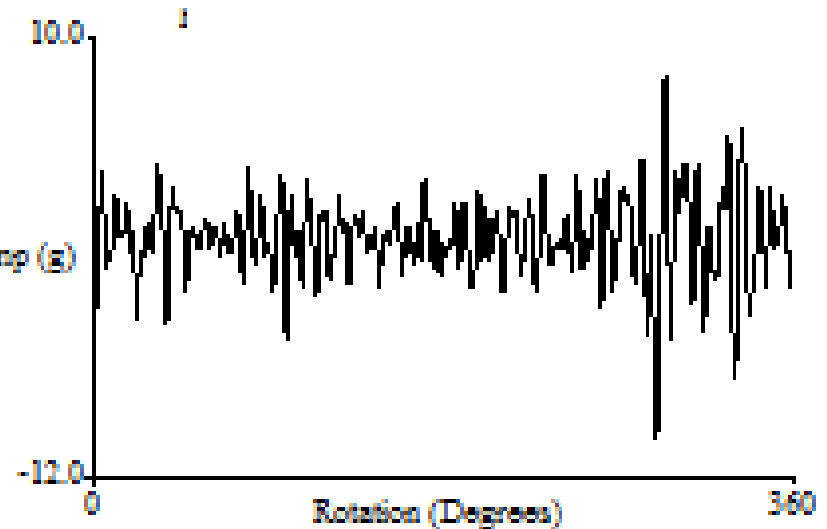
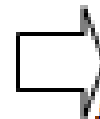
INPUT PINION

Wessex Input Pinion Failure

Subtraction of the 'normal' gear meshing signature from the synchronous average gives the 'error' or residual signal.



Synchronous Average
Cracked Wessex Input Pinion

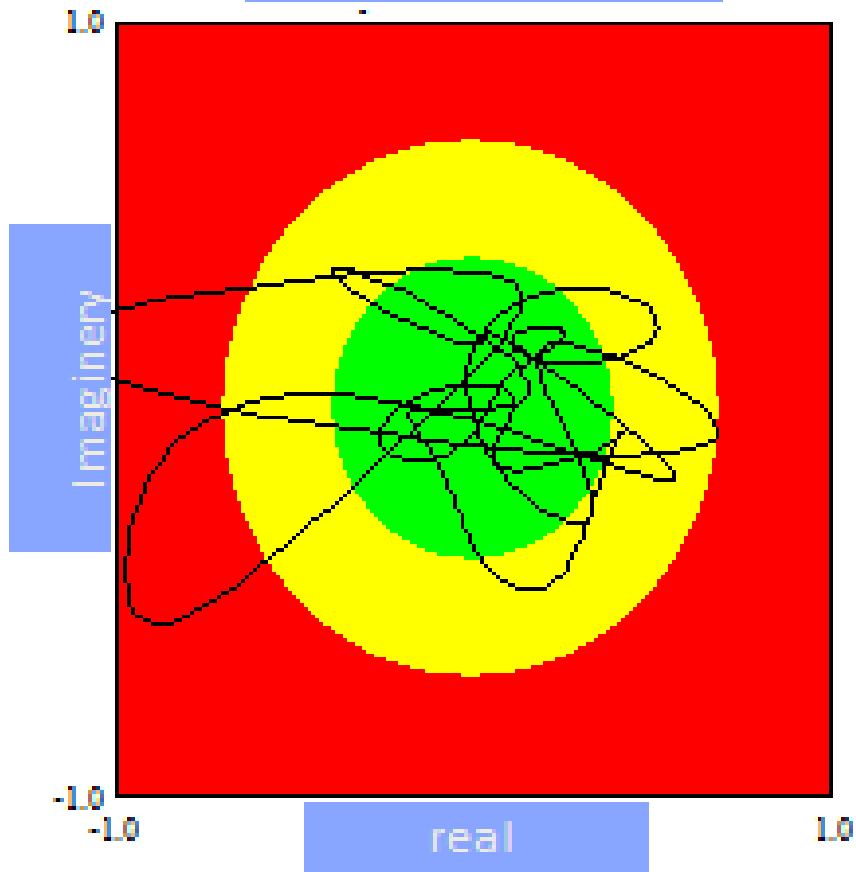
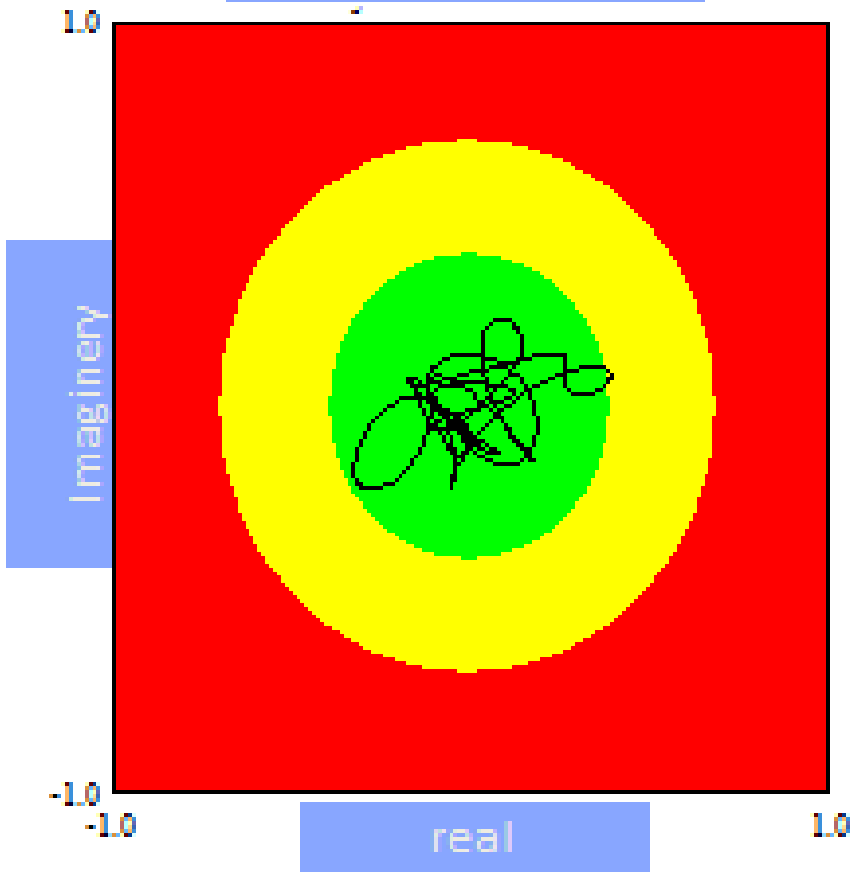


Residual Signal
Cracked Wessex Input Pinion

Residuals can be transformed into 'Bullseye Plots'

318 hrs before failure

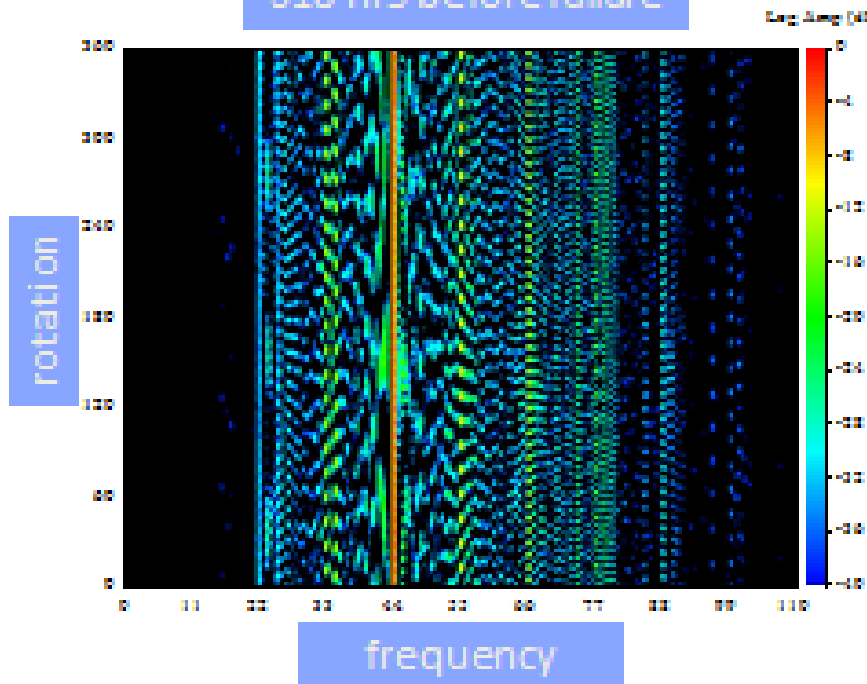
42 hrs before failure



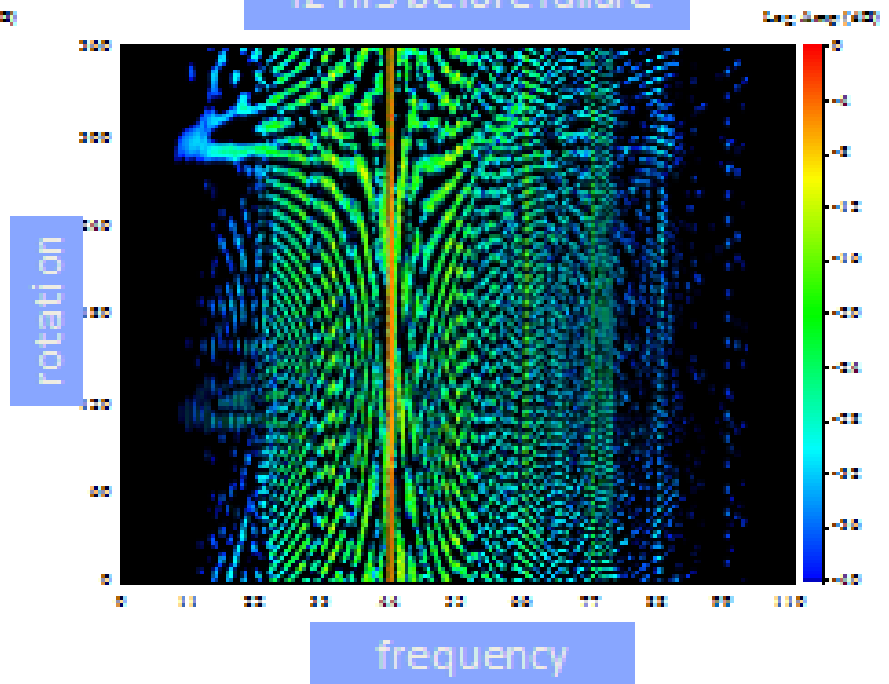
- includes phase information
- interpretation simplified

Advanced analyses retain more information of the meshing vibration

318 hrs before failure



42 hrs before failure



Wigner-Ville Distributions

- no filtering required
- contains much more information on fault

